# Dynamics and Relativity Summary

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## 1 Newtonian Dynamics -Basic Concepts

#### 1.1 Particles

**Note.** Velocity is tangent to path (or trajectory) (See Vector Calculus for more details)

**Definition.** The **momentum** of a particle is:

 $m\mathbf{u} = m\dot{\mathbf{r}} = \mathbf{p}$ 

(this notation is often used for momentum)

#### 1.2 Newton's Laws of Motion

**Law** (Newton's 1<sup>st</sup> Law). There exist inertial frames of reference (or inertial frames) in which a particle remains at rest or moves in a straight line at constant speed (i.e. it moves at constant velocity) unless it is acted on by a force. (Galileo's Law of Inertia)

Law (Newton's  $2^{nd}$  Law). In an inertial frame, the rate of change of momentum of a particle is equal to the force acting on it.

**Note.** This is a statement about vectors.

Law (Newton's  $3^{rd}$  Law). To every action there is an equal and opposite reaction. Forces exerted between two particles are equal in magnitude and opposite in direction.

**Remark.** All of these statements about particles can be extended to finite bodies (comprised of many particles).

## 1.3 Inertial frames & Galilean Transformations

In inertial frames, acceleration is zero if force is zero.

$$\ddot{\mathbf{r}} = \mathbf{0} \iff \mathbf{F} = \mathbf{0}$$

Inertial frames are not unique,

If S is an inertial frame, then any other frame S' moving with constant velocity relative to S is also inertial.



**Definition.** A **boost** is a transformation of form:

 $\mathbf{r}' = \mathbf{r} - \mathbf{v}t$ 

Where  $\mathbf{v}$  is velocity of S' relative to S. (Generalised arbitrary direction of above)

For a particle with position  $\mathbf{r}(t)$  in S and  $\mathbf{r}'(t')$  in S', then velocity  $\mathbf{u} = \dot{\mathbf{r}}(t)$  in S and acceleration  $\mathbf{a} = \ddot{\mathbf{r}}(t)$  in S relate to values in S' by:

$$\mathbf{u}' = \mathbf{u} - \mathbf{v}, \ \mathbf{a}' = \mathbf{a}$$

**Definition.** The **Galilean group** (group of Galilean transformations) is generated by the set of transformations that preserve inertial frames. Boosts combined with (some of) the following:

- translation of space:  $\mathbf{r}' = \mathbf{r} \mathbf{r}_0$ ,  $\mathbf{r}_0$  constant
- translation of time:  $t' = t t_0$
- rotations and reflections in space:  $\mathbf{r}' = R\mathbf{r}$  where R is an orthogonal matrix

**Note.** For any Galilean transformation we have

 $\ddot{\mathbf{r}} = \mathbf{0} \iff \ddot{\mathbf{r}}' = 0$ 

S inertial  $\iff S'$  inertial

#### 1.3.1 Galilean relativity

**Note.** Principle of Galilean relativity is that laws of Newtonian physics are the same in all inertial frames.

i.e. laws of physics look the same:

- at any point in space
- at any time
- in whatever direction I face
- whatever constant velocity I move with

Any set of equations which describe Newtonian physics must have Galilean invariance.

**Remark.** Measurement of velocity is not absolute but measurement of acceleration is absolute

#### 1.4 Newton's Second Law and Equations of Motion

From  $2^{nd}$  Law as stated previously, for a particle subject to a force **F**, the momentum **p** satisfies:

$$\frac{\mathrm{d}\mathbf{p}}{\mathrm{d}t} = \mathbf{F}$$
 where  $\mathbf{p} = m\mathbf{u} = m\dot{\mathbf{r}}$ 

Assume m is constant. (For variable mass see later in course.) Then

$$m\frac{\mathrm{d}\mathbf{u}}{\mathrm{d}t} = m\ddot{\mathbf{r}} = \mathbf{F}$$

Mass is a measure of 'reluctance to accelerate' i.e. inertia.

If **F** is specified as a function of **r**,  $\dot{\mathbf{r}}$ , t, i.e.  $\mathbf{F}(\mathbf{r}, \dot{\mathbf{r}}, t)$ , then we have a 2<sup>nd</sup> order differential equation for  $\mathbf{r}(t)$ , i.e.

$$m\ddot{\mathbf{r}} = m\frac{\mathrm{d}^{2}\mathbf{r}}{\mathrm{d}t^{2}} = \mathbf{F}\left(\mathbf{r}, \frac{\mathrm{d}\mathbf{r}}{\mathrm{d}t}, t\right)$$

Need to provide initial position  $\mathbf{r}(t_0)$ , inertial velocity  $\frac{d}{dt}\mathbf{r}(t_0)$  then have unique solution. The path/ trajectory of the particle is then determined (at all future times and at all past times).

## 2 Dimensional Analysis

Many problems in dynamics involve 3 basic dimensional quantities:

- L length
- $\bullet~M$  mass
- T time

dimensions of the same quantity [x] can be expressed in terms of L, M, T

$$\begin{array}{c|c} [\text{density}] & M \cdot L^{-3} \\ [\text{force}] & M \cdot L \cdot T^{-2} \end{array}$$

Only power law functions of M, L, T are allowed e.g. don't allow  $e^X = 1 + X + \frac{X^2}{2} + \dots$  with X dimensional and similarly complicated functions.

#### 2.1 Units

Introduce units for basic dimensional quantities, L, M, T e.g. SI Units

- m (metres) for length L
- kg (kilogrammes) for mass M
- sec (seconds) for time T

Many other physical quantities can be formed out of these basic units: e.g. G appearing in Newton's Law of Gravitation.

$$F = \frac{Gm_1m_2}{r^2}$$

Hence dimensions of G

$$G\sim \frac{Fr^2}{m_1m_2}=\frac{L^3}{MT^2}$$

Natural units for  $G: \text{ m}^3\text{kg}^{-1}\text{sec}^{-2}$ 

$$G = 6.67 \times 10^{-11} \,\mathrm{m^3 kg^{-1} sec^{-2}}$$

Note. General principle - dynamical/ physical equations must work for any consistent choice of units

#### 2.2 Scaling



## 3 Forces

#### 3.1 Force and Potential Energy in 1 Spatial Dimension

Consider mass m moving in a straight line with position x(t). Force depends only on position x. not on velocity  $\dot{x}$  or time t. Let F(x) be the force.

**Definition.** Define **potential energy** V(x) by

$$F(x) = -\frac{\mathrm{d}V}{\mathrm{d}x}$$

Hence

$$V(x) = -\int^x F(x') \,\mathrm{d}x'$$

Lower limit omitted  $\implies$  arbitrary constant in V.

**Equation.** Equation of motion determined by Newton's  $2^{nd}$  law

$$m\ddot{x} = -\frac{\mathrm{d}V}{\mathrm{d}x}$$

Definition. Define kinetic energy

$$T = \frac{1}{2}m\dot{x}^2$$

(will generalise to  $T = \frac{1}{2}m|\dot{\mathbf{x}}|^2$  in more than 1 dimension)

Equation. Total energy

$$E = T + V = \frac{1}{2}m\dot{x}^2 + V(x)$$

Claim. Total energy is conserved i.e.

$$\frac{\mathrm{d}E}{\mathrm{d}t} = 0$$

**Proof.** Trivial (plug in for E)

Method. Determining x(t) given potential.

In 1-D, conservation of energy gives useful information about the motion. Conservation of energy is a 1<sup>st</sup> integral of Newton's 2<sup>nd</sup> law.

$$E = \frac{1}{2}m\dot{x}^2 + V(x)$$

is constant. Hence, can rearrange for  $\dot{x}$  and integrate wrt t to get x(t), or take reciprocal and integrate wrt x to get t(x)

**Example** (Qualitative insight from conservation of energy).

$$V(x) = \lambda(x^3 - 3\beta^2 x) \ \lambda, \beta \text{ positive constants}$$

What happens if we release particle from rest at  $x = x_0$  for different choices of  $x_0$ ?

- $E = V(x_0)$  in subsequent motion  $V(x) \leq V(x_0)$  as kinetic energy positive.
  - Case 1:  $x_0 < -\beta$  particle moves to left with  $x(t) \to -\infty$  as  $t \to \infty$
  - Case 2:  $-\beta < x_0 < 2\beta$  particle remains confined to  $-\beta < x(t) < 2\beta$
  - Case 3:  $2\beta < x_0$  particle moves to left, reaches  $x = -\beta$  and continues with  $x(t) \to -\infty$  as  $t \to \infty$ .
  - Special case 1:  $x_0 = -\beta$  particle stays at  $x = -\beta$
  - Special case 2:  $x_0 = \beta$  particle stays at  $x = \beta$

• Special case 3:  $x_0 = 2\beta$  particle moves to left and comes to rest at  $x = -\beta$ 

For special case 3, write down integral expression relating x and t.

$$\begin{split} t &= \int_{x(t)}^{2\beta} \frac{\mathrm{d}\tilde{x}}{\sqrt{\frac{2\lambda}{m} (2\beta^3 - \tilde{x}^3 + 3\beta^2 \tilde{x})}} \\ &= \int_{x(t)}^{2\beta} \frac{\mathrm{d}\tilde{x}}{\sqrt{\frac{2\lambda}{m} (\tilde{x} + \beta)^2 (2\beta - \tilde{x})}} \\ &= \int_{x(t)}^{2\beta} \frac{1}{\sqrt{\frac{2\lambda}{m}}} \cdot \frac{1}{\tilde{x} + \beta} \cdot \frac{1}{(2\beta - \tilde{x})^{1/2}} \,\mathrm{d}\tilde{x} \text{ (this diverges as } \tilde{x} \to -\beta) \end{split}$$

particle takes infinite time to reach  $x = -\beta$ . (logarithmic behaviour)

#### 3.2 Equilibrium Points

**Definition.** An equilibrium point is a point at which a particle can stay at rest for all time. In the previous example,  $x = \pm \beta$  are equilibrium points. General condition: V'(x) = 0 at equilibrium. **Method.** Analyse motion near equilibrium at  $x = x_0$ , so  $V'(x_0) = 0$ , assume that  $x - x_0$  is small, expand V(x) in Taylor Series: V''(x) gives information on maximum (unstable) or minimum (stable)

Equation (Pendulum).

$$P\simeq 2\pi \left(\frac{l}{g}\right)^{1/2}$$

#### 3.3 Force and potential for motion in 3-D

**Equation.** Consider a particle moving in 3-D under force  $\mathbf{F}$ 

Total work = 
$$\int_{t_1}^{t_2} \mathbf{F} \cdot d\mathbf{r} = \int_{t_1}^{t_2} F_x dx + F_y dy + F_z dz$$
  
 $\mathbf{F} = (F_x, F_y, F_z)$ 

**Definition.** Now suppose  $\mathbf{F}$  is a function of  $\mathbf{r}$  only.  $\mathbf{F}(\mathbf{r})$  defines a 'force field'. A conservative force field is such that  $\mathbf{F}(\mathbf{r}) = -\nabla V(\mathbf{r})$  for some function  $V(\mathbf{r})$ 

**Note.** In components  $F_i = - \frac{\partial V}{\partial x_i}$ 

**Claim.** If **F** is conservative then the energy  $E = T + V(\mathbf{r})$  is conserved.

**Proof.** Use

$$\frac{\mathrm{d}}{\mathrm{d}t}V(\mathbf{r}) = \nabla V\cdot\dot{\mathbf{r}}$$

**Equation.** Total work done by a conservative force:

$$W = V(\mathbf{r}_1) - V(\mathbf{r}_2)$$

(follows from properties of  $\nabla$ ) This is independent of the path taken between  $\mathbf{r}_1$  and  $\mathbf{r}_2$ .

**Corollary.** If C is closed then no work is done by force.

#### 3.4 Gravity

We have already noted the gravitational force felt by a mass m due to a mass M

$$\mathbf{F}(\mathbf{r}) = -\frac{GMm}{|\mathbf{r}|^3}\mathbf{r} = -\frac{GMm}{r^2}\hat{\mathbf{r}}$$

 ${\bf r}$  is position vector of mass m relative to mass M

 $\mathbf{F}(\mathbf{r}) = -\nabla V$ 

with

$$V(\mathbf{r}) = -\frac{GMm}{r}$$

Definition. We often define a "gravitational potential"

$$\Phi_g(\mathbf{r}) = -\frac{GM}{r}$$

and "gravitational field"

$$\mathbf{g} = -
abla \Phi_g(\mathbf{r}) = -rac{GM}{r^2}\hat{\mathbf{r}}$$

**Equation.** These are properties of mass M alone. Effect on mass m:

$$V(\mathbf{r}) = m\Phi_g(\mathbf{r}), \ \mathbf{F}(\mathbf{r}) = m\mathbf{g}$$

Remark. We can generalise to gravitational potential associated with many masses

$$M_i \ i=1,\ldots,n$$

and position vectors

$$\mathbf{r}_i \ i=1,\ldots,n$$

Equation. Many particles:

$$\Phi_g(\mathbf{r}) = -\sum_{i=1}^{N} \frac{GM_i}{|\mathbf{r} - \mathbf{r}_i|}$$
$$\mathbf{g} = -\sum_{i=1}^{N} \frac{GM_i}{|\mathbf{r} - \mathbf{r}_i|^3} (\mathbf{r} - \mathbf{r}_i)$$

Could extend to a continuous distribution of mass by generalising sums to integrals. In particular, for a uniform spherical mass distribution centered at origin, we have

$$\Phi_g(\mathbf{r}) = -\frac{GM}{r}$$

with M the total mass – equivalent to point mass at origin (see VC)

#### 3.4.1 Simple Results on gravity

(i) 1-D approximation: Use

$$V(R+z) = -\frac{GMm}{R+z} = -\frac{GMm}{R} \cdot \frac{1}{1+z/R}$$

and expand for z

(ii) Escape Velocity:

Consider leaving surface of a planet with velocity v. Conservation of energy:

$$E = T + V = \frac{1}{2}mv^2 - \frac{GM}{r}$$

Set total energy = 0 to get

$$v \ge \sqrt{\frac{2GM}{R}} =$$
escape velocity

#### 3.5 Electromagnetic Forces

**Equation.** We have that force  $\mathbf{F}$  acting on a particle with charge q is

$$\mathbf{F} = q\mathbf{E} + q\mathbf{u} \times \mathbf{B}$$
 where  $\mathbf{u} = \dot{\mathbf{r}}$ 

(Lorentz force law) with  $\mathbf{E}(\mathbf{r}, t)$  electric field and  $\mathbf{B}(\mathbf{r}, t)$  magnetic field. Restrict to time independent fields  $\mathbf{E}(\mathbf{r}), \mathbf{B}(\mathbf{r})$ . In this case we can write

$$\mathbf{E} = -\nabla \Phi_e(\mathbf{r})$$

 $\Phi_e$  is the "electrostatic potential". The force  $q{\bf E}$  is then conservative.

Claim. For time independent  $\mathbf{E}(\mathbf{r})$ ,  $\mathbf{B}(\mathbf{r})$  the energy of a particle moving under the Lorentz force law is constant

**Proof.** Prove in same way as general case

Equation. A particle with charge Q located at origin generates an electrostatic potential

$$\Phi_e(\mathbf{r}) = \frac{Q}{4\pi\varepsilon_0 r}$$

Where r is distance from origin to **r**. Electric field:

$$\mathbf{E}(\mathbf{r}) = -\nabla \Phi_e = \frac{Q}{4\pi\varepsilon_0 r^2} \hat{\mathbf{r}}$$

 $\varepsilon_0$  is a constant

$$8.85 \times 10^{-12} \,\mathrm{m}^{-3} \mathrm{kg}^{-1} \mathrm{s}^{2} \mathrm{C}^{2}$$

"electric constant" (C is the unit of charge)

**Equation.** Force on particle with charge q located at  $\mathbf{r}$ 

$$\mathbf{F} = \frac{Qq}{4\pi\varepsilon_0 r^2}\hat{\mathbf{r}}$$

'Coulomb force'

#### 3.6 Friction

Definition. Dry friction – solid bodies in contact





Definition. Kinetic friction: if block starts to slide, there is a kinetic frictional force

Example.

$$|\mathbf{F}| = \mu_k |\mathbf{N}|$$

 $\mu_k$  is coefficient of kinetic friction Expect  $\mu_s > \mu_k > 4$ 

**Definition.** Fluid drag – solid body moving through a fluid experiences a drag force.

Two important regimes for drag:

Equation. Linear drag:

 $\mathbf{F} = -k_1 \mathbf{u}$ 

where k constant and **u** is velocity of body relative to fluid,

**Remark.** This is relevant to 'small' objects moving through a viscous fluid (e.g. bacterium moving through body fluid)

Law (Stokes Drag Law).

 $k_1 = 6\pi\eta R$ 

for a moving sphere.  $\eta$  is viscosity of fluid, R is raius of sphere

Equation. Quadratic drag: large bodies moving through a less viscous fluid

 $\mathbf{F} = -k_2 |\mathbf{u}| \mathbf{u}$ 

with  $k_2$  different to  $k_1$ . Typically,

 $k_2 = \rho_{\rm fluid} C_D R^2$ 

where  $\rho_{\text{fluid}}$  is fluid density,  $C_D$  drag coefficient,  $R^2$  is size Relevant to swimming of large fish, cars, aircraft, ...

#### 3.7 Angular Momentum

In previous 3 subsections, we focused on specific types of force – gravitational, electromagnetic, frictional.

Now to conclude section 3, we return to more general aspects of the dynamics of a single particle.

**Definition.** The **angular momentum** for a particle of mass m moving under influence of a force **F**, position vector  $\mathbf{r}(t)$ , velocity  $\dot{\mathbf{r}}(t)$  is



Equation. Then

$$\frac{\mathrm{d}\mathbf{L}}{\mathrm{d}t} = m\dot{\mathbf{r}} \times \dot{\mathbf{r}} + m\mathbf{r} \times \ddot{\mathbf{r}} = \mathbf{r} \times \mathbf{F} = \mathbf{G}$$

Definition. We say  ${\bf G}$  above is the 'torque'

**Remark.** The values of **L** and **G** depend on choice of O - 'about the origin' or 'about ' any specified point.

Note. If  $\mathbf{r} \times \mathbf{F} = \mathbf{0}$  or equivalently  $\mathbf{G} = \mathbf{0}$  then  $\mathbf{L}$  is a constant vector i.e. angular momentum is conserved.

Angular momentum about some suitably chosen point may be constant

## 4 Orbits

The basic problem:

$$m\ddot{\mathbf{r}} = -\nabla V(r)$$

Particle moving in a force that is associated with potential that is only a function of radius – force is directed towards (or away from) the origin.

We are assuming that 'central' mass is remaining fixed – good approximation if the central mass is much larger than m (we relax this assumption later in the course)

#### 4.1 Central Forces

Definition. Cent	ral forces are a	special class of o	conservative forces with
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$$V(\mathbf{r}) = V(|\mathbf{r}|)$$

which gives

$$\mathbf{F}(\mathbf{r}) = -\frac{\mathrm{d}V}{\mathrm{d}r}\hat{\mathbf{r}}$$

and we can check:

$$\implies \nabla r = \frac{\mathbf{x}}{r} = \mathbf{i}$$

**Remark.** Have  $\mathbf{L} = \text{constant}$ , also  $\mathbf{L} \cdot \mathbf{r} = 0$  (from definition of  $\mathbf{L}$ ) Hence motion is in a plane through the origin O with orientation set by value of  $\mathbf{L}$ 

#### 4.2 Polar Co-Ordinates in the Plane

**Definition.** We define unit vectors: in direction of increasing r:  $\mathbf{e}_r$ in direction of increasing  $\theta$ :  $\mathbf{e}_{\theta}$  $\mathbf{e}_r = \hat{\mathbf{r}} = \begin{bmatrix} \cos \theta \\ \sin \theta \end{bmatrix} \mathbf{e}_{\theta} = \begin{bmatrix} -\sin \theta \\ \cos \theta \end{bmatrix}$ Relative to fixed Cartesian axes At any point  $\mathbf{e}_r, \mathbf{e}_{\theta}$  form an orthonormal basis, but orientation depends on position. Note.

$$rac{\mathrm{d}}{\mathrm{d} heta}\mathbf{e}_r = \mathbf{e}_ heta$$
 $rac{\mathrm{d}}{\mathrm{d} heta}\mathbf{e}_ heta = -\mathbf{e}_t$ 

Use chain rule for derivative wrt time

Equation.

$$\dot{\mathbf{r}} = \dot{r}\mathbf{e}_r + r\dot{\theta}\mathbf{e}_\theta$$

The acceleration:

$$\ddot{\mathbf{r}} = (\ddot{r} - r\dot{\theta}^2)\mathbf{e}_r + (2\dot{r}\dot{\theta} + r\ddot{\theta})\mathbf{e}_{\theta}$$

**Example.** Circular motion: Velocity  $\dot{\mathbf{r}} = a\omega \mathbf{e}_{\theta}$ Acceleration  $\ddot{\mathbf{r}} = -a\omega^2 \mathbf{e}_r$ 

## 4.3 Motion in a Central Force Field

## Method.

$$r^2\dot{\theta} = h \text{ (constant)}$$

(comes from  $\mathbf{L}$ ) constant. Radial component of force:

$$m\ddot{r} - mr\dot{\theta}^2 = \frac{\mathrm{d}V}{\mathrm{d}r}$$
$$m\ddot{r} = -\frac{\mathrm{d}V_{\mathrm{eff}}}{\mathrm{d}r}$$

With

$$V_{\rm eff}(r) = V(r) + rac{1}{2} rac{mh^2}{r^2}$$

effective potential. i.e. motion particle is equivalent to 1-D motion under the influence of the effective potential. We are just focusing on radius here.

Consider energy of particle:

$$T + V(r) = \frac{1}{2}m\dot{r}^{2} + \underbrace{\frac{1}{2}m\frac{h^{2}}{r^{2}} + V(r)}_{V_{\text{eff}}(r)}$$

**Example** (inverse square law force).

What is the possible motion of m?

If  $E_{\min} < E < 0$ , r(t) oscillates between minimum and maximum values,  $\dot{\theta}$  varies. If  $0 \le E$ ,  $r(t) \to \infty$  as  $t \to \infty$ , the particle escapes (unbound orbit)

**Note.**  $r_{\min}$  is called the periapsis, perihelion for Sun, perigee for Earth  $r_{\max}$  is called the apoasis, aphelion for Sun, apogee for Earth

#### 4.4 Stability of Circular Orbits

Method. Consider a general potential V(r). Does a circular orbit exist and is it stable? Assume given h where  $h \neq 0$ . Need:  $V'_{\text{eff}}(r_*) = 0$ 

and stable iff

$$V''(r_*) + \frac{3V'(r_*)}{r_*} > 0$$

(Comes from  $V_{\text{eff}}''(r_*) < 0$ )

## 4.5 The Orbit Equation

**Method.** To determine shape of orbit, use  $\theta$  as the independent variable, by writing

$$\frac{\mathrm{d}}{\mathrm{d}t} = \frac{\mathrm{d}\theta}{\mathrm{d}t} \cdot \frac{\mathrm{d}}{\mathrm{d}\theta} = \frac{h}{r^2} \frac{\mathrm{d}}{\mathrm{d}\theta}$$

apply to Netwon's 2<sup>nd</sup> Law:

$$F(r) = m\frac{h}{r^2}\frac{\mathrm{d}}{\mathrm{d}\theta}\left(\frac{h}{r^2}\frac{\mathrm{d}}{\mathrm{d}\theta}r\right) - \frac{mh^2}{r^3}$$

Subbing u = 1/r yields:

$$mhu^2 \frac{\mathrm{d}}{\mathrm{d}\theta} \left( -h\frac{\mathrm{d}u}{\mathrm{d}\theta} \right) - mh^2 u^3 = F(u^{-1})$$

which rearranges to give:

Equation (Orbit equation, or Binet equation).

$$\frac{\mathrm{d}^2 u}{\mathrm{d}\theta^2} + u = -\frac{1}{mh^2 u^2} F(u^{-1})$$

Solve this to find  $u(\theta)$ , then use  $\dot{\theta} = hu^2$  etc.

## 4.6 The Kepler Problem

Method. This is the orbit problem for the special case of a gravitational central force.

$$F(r) = -\frac{mk}{r^2}$$

Solving the orbit equation gives:

Equation.

$$r = \frac{1}{u} = \frac{l}{1 + e \cos \theta}$$
$$l = \frac{h^2}{k}$$
$$e = \frac{Ah^2}{k}$$

**Method.** If  $0 \le e < 1$ : ellipse - orbit is bounded

$$\frac{l}{1+e} \le r(\theta) \le \frac{l}{1-e}$$
Rewrite (†)  
with  

$$\frac{(x+ea)^2}{a^2} + \frac{y^2}{b^2} = 1$$

$$a = \frac{l}{1-e^2}$$

$$b = \frac{l}{\sqrt{1-e^2}}$$

Method. e = 0: circle

a = b = radius of circle

e>0 - orgin lies in one of the foic of the ellipse





**Method.** e > 1: Hyperbola  $r \to \infty$  as  $\theta \to \pm \alpha$  with

$$\alpha = \cos^{-1}(-\frac{1}{e}) \in \left(\frac{\pi}{2}, \pi\right)$$

Rewrite  $(\dagger)$  as

$$\frac{(x-ea)^2}{a^2} - \frac{y^2}{b^2} = 1$$
$$a = \frac{l}{e^2 - 1}$$
$$b = \frac{l}{\sqrt{e^2 - 1}}$$

with

Hyperbolic orbit represents incoming body with large velocity which is deflected by gravitational force.



$$\mathbf{r} \cdot \mathbf{n} = \begin{bmatrix} x \\ y \end{bmatrix} \cdot \frac{1}{\sqrt{a^2 + b^2}} \begin{bmatrix} b \\ \mp a \end{bmatrix} = \frac{bx \mp ay}{\sqrt{a^2 + b^2}} = \frac{eba}{\sqrt{a^2 + b^2}} = b$$

(important parameter)

Method. e = 1: parabola with equation

$$r = \frac{l}{1 + \cos \theta}$$

 $r \to \infty$  as  $\theta \to \pm \pi$ In Cartesians:

$$y^2 = l(l - 2x)$$

Marginal case between ellipse and hyperbola.

#### 4.6.1 Energy and Eccentricity

Equation. Recall:

$$E = \frac{mk}{2l}(e^2 - 1)$$

Using  $\dot{r} = -h \frac{du}{d\theta}$  and  $l = \frac{h^2}{k}$ Bound orbits have e < 1, E < 0 and unbound orbits have e > 1, E > 0. Marginal case e = 1, E = 0note also

$$e = \left(\frac{2lE}{mk} + 1\right)^{1/2}$$

#### 4.6.2 Keplers Laws of Planetary Motion

Law (Keplers Laws of Planetary Motion). (i) Orbit of planet is ellipse with Sun at focus (ii) Line between planet and Sun sweeps out equal area in equal times (iii) Square of period P is proportional to cube of semi-major axis a $P^2 \propto a^3$ (i) is consistent with our solution of orbit equation (for bounded orbits) (ii) O $\delta\theta = \text{small change in } \theta \text{ in time } \delta t$ area  $\simeq \frac{1}{2}r^2 d\theta$ Hence rate of change is  $\frac{1}{2}r^2\dot{\theta}$  = angular momentum h i.e. follows from conservation of angular momentum (iii) Note area of ellipse is  $\pi ab = \frac{h}{2}P$ recall  $b^2 = a^2(1 - e^2)$  $h^2 = kl = ka(1 - e^2)$ Now consider  $P^2 = \frac{4\pi^2 a^3}{k} \propto a^3$ by subbing in for  $b^2$  and  $h^2$ Note. The multiplying constant is same for all masses orbiting central mass M

## 5 Rotating Frames of Reference

**Note.** Newton's Laws valid are only in an inertial frame of reference. A rotating frame is non-inertial and therefore the equation of motion will be modified relative to Newtons  $2^{nd}$  Law.

**Method.** Let S be an inertial frame S' be a non-inertial frame – rotating about z in S – with angular velocity  $\omega (= \dot{\theta} \text{ where } \theta \text{ is angle between } x \text{ or } y \text{ axis in } S \text{ and that in } S')$ 



for S'

$$\mathbf{e}'_i = \{\hat{\mathbf{x}'}, \ \hat{\mathbf{y}'}, \ \hat{\mathbf{z}'}\} = \{\mathbf{e}'_1, \ \mathbf{e}'_2, \ \mathbf{e}'_3\}$$

Consider a particle at rest in S'. viewed in S its velocity:

$$\left(\frac{\mathrm{d}\mathbf{r}}{\mathrm{d}t}\right)_{S} = \boldsymbol{\omega} \times \mathbf{r} \text{ where } \boldsymbol{\omega} = \omega \hat{\mathbf{z}}$$

Note. Angular velocity vector is aligned with the axis of rotation, magnitude is equal to scalar angular velocity, and viewed from the direction of the vector, the rotation is anticlockwise if  $\omega > 0$ 

**Equation.** Same formula applies to any vector that is fixed in S'; in particular to the basis vectors  $\mathbf{e}'_i$  i.e.

$$\left(\frac{\mathrm{d}}{\mathrm{d}t}\mathbf{e}_i\right)_S = \boldsymbol{\omega} \times \mathbf{e}'_i$$

(note that  $\frac{d}{dt}\mathbf{e}'_z = 0$  under rotation assumed here)

Method. Consider a general time dependent vector  ${\bf a}$ 

$$\mathbf{a}(t) = \sum_{i=1}^{3} a'_i(t) \mathbf{e}'_i(t)$$

expression of **a** in terms of components defined in S'Now consider rate of change of  $\mathbf{a}(t)$ 

$$\left(\frac{\mathrm{d}}{\mathrm{d}t}\mathbf{a}(t)\right)_{S'} = \sum_{i=1}^{3} \left(\frac{\mathrm{d}}{\mathrm{d}t}a'_{i}(t)\right)\mathbf{e}_{i}(t)$$

gives rate of change observed in S'. What about rate of change observed in S?

Equation.

$$\begin{pmatrix} \frac{\mathrm{d}}{\mathrm{d}t} \mathbf{a}(t) \end{pmatrix}_{S} = \sum_{i=1}^{3} \left( \frac{\mathrm{d}}{\mathrm{d}t} a_{i}'(t) \right) \mathbf{e}_{i}'(t) + \sum_{i=1}^{3} a_{i}'(t) \frac{\mathrm{d}}{\mathrm{d}t} \mathbf{e}_{i}'(t)$$
$$\Rightarrow \left( \frac{\mathrm{d}}{\mathrm{d}t} \mathbf{a}(t) \right)_{S} = \left( \frac{\mathrm{d}}{\mathrm{d}t} \mathbf{a} \right)_{S'} + \boldsymbol{\omega} \times \mathbf{a}$$

**Remark.** This is a key identity which relates rate of change of vectors seen in S' to rate of change seen in S.

**Example.** Apply to position vector  ${\bf r}$ 

$$\left(\frac{\mathrm{d}\mathbf{r}}{\mathrm{d}t}\right)_{S} = \left(\frac{\mathrm{d}\mathbf{r}}{\mathrm{d}t}\right)_{S'} + \boldsymbol{\omega} \times \mathbf{r} \text{ (velocity)}$$

Note that the difference depends on position. Now apply to velocity - allow  $\boldsymbol{\omega}$  to depend on time

$$\begin{pmatrix} \frac{\mathrm{d}^{2}\mathbf{r}}{\mathrm{d}t^{2}} \end{pmatrix}_{S} = \left( \left(\frac{\mathrm{d}}{\mathrm{d}t}\right)_{S'} + \boldsymbol{\omega} \times \right) \left( \left(\frac{\mathrm{d}}{\mathrm{d}t}\right)_{S'} + \boldsymbol{\omega} \times \right) \mathbf{r}$$

$$= \left( \left(\frac{\mathrm{d}}{\mathrm{d}t}\right)_{S'} + \boldsymbol{\omega} \times \right) \left( \left(\frac{\mathrm{d}\mathbf{r}}{\mathrm{d}t}\right)_{S'} + \boldsymbol{\omega} \times \mathbf{r} \right)$$

$$= \left(\frac{\mathrm{d}^{2}\mathbf{r}}{\mathrm{d}t^{2}}\right)_{S'} + 2\boldsymbol{\omega} \times \left(\frac{\mathrm{d}\mathbf{r}}{\mathrm{d}t}\right)_{S'} + \dot{\boldsymbol{\omega}} \times \mathbf{r} + \boldsymbol{\omega} \times (\boldsymbol{\omega} \times \mathbf{r})$$

Equation. Equation of motion in a rotating frame

$$n\left(\frac{\mathrm{d}^{2}\mathbf{r}}{\mathrm{d}t^{2}}\right)_{S} = \mathbf{F}$$

$$= m\left[\left(\frac{\mathrm{d}^{2}\mathbf{r}}{\mathrm{d}t^{2}}\right)_{S'} + \underbrace{2\boldsymbol{\omega}\times\left(\frac{\mathrm{d}\mathbf{r}}{\mathrm{d}t}\right)_{S'} + \dot{\boldsymbol{\omega}}\times\mathbf{r} + \boldsymbol{\omega}\times\left(\boldsymbol{\omega}\times\mathbf{r}\right)}_{\text{'fictitious forces'}}\right]$$

**Note.** Need to take account of fictitious forces to explain motion observed in the rotating frame (or more general non-inertial)

Coriolis force:

$$-2m\dot{\boldsymbol{\omega}} imes \left(\frac{\mathrm{d}\mathbf{r}}{\mathrm{d}t}\right)_{s}$$

Euler force:

 $-m\dot{\boldsymbol{\omega}} imes \mathbf{r}$ 

(in many applications take this to be zero) Centrifugal force

 $-m\boldsymbol{\omega} \times (\boldsymbol{\omega} \times \mathbf{r})$ 

#### 5.1 Centrifugal Force



$$n\omega^2 \mathbf{r}_{\perp} = \nabla \left( \frac{1}{2} m \omega^2 |\mathbf{r}_{\perp}|^2 \right)$$

(can show)

i.e. centrifugal force is a conservative force. On a rotating planet, it is convenient to combine centrifugal force and gravitational force into 'effective gravity'

Equation.

 $\mathbf{g}_{\mathrm{eff}} = \mathbf{g} + \omega^2 \mathbf{r}_{\perp}$ 



## 5.2 Coriolis force

Method. Using  $-2m\boldsymbol{\omega} \times \left(\frac{\mathrm{d}\mathbf{r}}{\mathrm{d}t}\right)_{S'} = -2m\boldsymbol{\omega} \times \mathbf{v}$ (v shorthand for velocity observed in rotating frame) we get: The horizontal Coriolis force gives an acceleration to the right of the horizontal velocity on the Northern Hemisphere, to the left in the Southern Henisphere

## 6 Systems of Particles

#### 6.1 Center of Mass

Note. Total mass

$$M = \sum_{i=1}^{N} m_i$$

Equation. Center of mass located at

$$\mathbf{R} = \frac{1}{M} \sum_{i=1}^{N} m_i \mathbf{r}_i$$

Equation. Total linear momentum

$$\mathbf{P} = \sum_{i=1}^{N} m_i \dot{\mathbf{r}}_i = \sum_{i=1}^{N} \mathbf{p}_i = M \dot{\mathbf{R}}$$

i.e. total linear momentum is equivalent to that of point mass M located at  $\mathbf{R}$ .

Equation. Then

$$\mathbf{P} = M\mathbf{R}$$
$$= \mathbf{F}^{\text{ext}}$$

(can show)

Center of mass moves as if it is the position of mass M under the influence of force

$$\mathbf{F}^{\mathrm{ext}} = \sum_{i=1}^{N} \mathbf{F}^{\mathrm{ext}}_{i}$$

Note. If  $\mathbf{F}^{\text{ext}} = 0$  then  $\dot{\mathbf{P}} = \mathbf{0}$  so total momentum is conserved. There will be a 'center of mass' frame with origin at center of mass – which is inertial. In this frame  $\dot{\mathbf{R}} = \mathbf{0}$ , e.g. take  $\mathbf{R} = 0$ 

#### 6.2 Motion Relative to Centre of Mass

 $\mathbf{r}_{i} = \mathbf{R} + \mathbf{s}_{i}$ where  $\mathbf{s}_{i}$  is position vector relative to the centre of mass  $\sum_{i=1}^{N} m_{i} \mathbf{s}_{i} = \mathbf{0}$ (can show)
From this we also get  $\frac{\mathrm{d}}{\mathrm{d}t} \sum_{i=1}^{N} m_{i} \mathbf{s}_{i} = \mathbf{0}$ 

Equation. Total linear momentum

$$\mathbf{P} = M\mathbf{R}$$

(can show)

Equation. Let

#### 6.3 Angular Momentum

Equation. Total angular momentum

$$\mathbf{L} = \sum_{i=1}^{N} \mathbf{r}_i \times \mathbf{p}_i \text{ (about } O)$$

$$\dot{\mathbf{L}} = \sum_{i=1}^{N} \mathbf{r}_{i} \times \mathbf{F}_{i}^{\text{ext}} + \frac{1}{2} \sum_{i=1}^{N} \sum_{j=1}^{n} (\mathbf{r}_{i} - \mathbf{r}_{j}) \times \mathbf{F}_{ij}$$

$$(= 0 \text{ if } \mathbf{F}_{ij} \parallel \mathbf{r}_{i} - \mathbf{r}_{j}) \qquad (*)$$

If (\*) is satisfied

$$\dot{\mathbf{L}} = \sum_{i=1}^{N} \mathbf{r}_i imes \mathbf{F}_i^{ ext{ext}} = \mathbf{G}^{ ext{ext}}$$

This is the total external torque acting on system

**Equation.** Now return to motion, position relative to center of mass Total angular momentum

$$\mathbf{L} = M(\mathbf{R} \times \dot{\mathbf{R}}) + \sum_{i=1}^{N} m_i \mathbf{s}_i \times \dot{\mathbf{s}}_i$$

So  $\mathbf{L}$  = angular momentum of a particle of mass M at  $\mathbf{R}$  moving with velocity  $\dot{\mathbf{R}}$  + angular momentum associated with motion of particles relative to the centre of mass

## 6.4 Energy

Equation. Total kinetic energy

$$T = \frac{1}{2}M\dot{\mathbf{R}}^2 + \frac{1}{2}M\dot{\mathbf{s}}_i^2$$

So  $T={\rm KE}$  of particle mass M moving with velocity  $\dot{R}+{\rm KE}$  associated with particle motion relative to the centre of mass

Now ask is energy conserved?

Method. Consider

$$\frac{\mathrm{d}T}{\mathrm{d}t} = \sum_{i=1}^{N} \dot{\mathbf{r}}_{i} \cdot \mathbf{F}_{i}^{\mathrm{ext}} + \sum_{i=1}^{N} \sum_{j>i}^{N} (\dot{\mathbf{r}}_{i} - \dot{\mathbf{r}}_{j}) \cdot \mathbf{F}_{ij}$$

If external forces are defined by a potential

$$\mathbf{F}_{i}^{\mathrm{ext}} = -\nabla_{\mathbf{r}_{i}} V_{i}^{\mathrm{ext}}$$

and internal forces are defined by a potential

$$\mathbf{F}_{ij} = -\nabla_{\mathbf{r}_i - \mathbf{r}_j} V_{ij} (\mathbf{r}_i - \mathbf{r}_j)$$

then

$$\frac{\mathrm{d}T}{\mathrm{d}t} = -\frac{\mathrm{d}}{\mathrm{d}t} \sum_{i=1}^{N} V^{\mathrm{ext}}(\mathbf{r}_i) - \frac{\mathrm{d}}{\mathrm{d}t} \sum_{i=1}^{N} \sum_{j>i}^{N} V(\mathbf{r}_i - \mathbf{r}_j)$$

i.e. we have a conservation of energy (move RHS terms to LHS)

## 6.5 Two-Body Problem



$$\mathbf{L} = M\mathbf{R} \times \dot{\mathbf{R}} + \mu \mathbf{r} \times \dot{\mathbf{r}}$$

(special forms for 2-body problem of general expressions derived earlier)

## 6.6 Variable Mass Problems

#### 6.6.1 Rocket Problem



Have a rocket moving in 1-dimension with speed v(t) and mass m(t). Rocket propels itself by expelling mass at velocity u relative to rocket

Total momentum is conserved because there are no external forces acting.

**Equation.** Total momentum at  $t + \delta t$ :

$$m(t+\delta t)v(t+\delta t) + (m(t) - m(t+\delta t))(v(t) - u + O(\delta t))$$

Change in momentum from t to  $t + \delta t$ :

$$m(t+\delta t)v(t+\delta t) + (m(t) - m(t+\delta t))(v(t) - u + O(\delta t)) - m(t)v(t)$$
$$\simeq \left(\frac{\mathrm{d}m}{\mathrm{d}t}u + m\frac{\mathrm{d}v}{\mathrm{d}t}\right)\delta t + O(\delta t^2) = 0$$
$$\implies \frac{\mathrm{d}m}{\mathrm{d}t}u + m\frac{\mathrm{d}v}{\mathrm{d}t} = 0$$

Generalise to

$$\frac{\mathrm{d}m}{\mathrm{d}t}u + m\frac{\mathrm{d}v}{\mathrm{d}t} = F_{\mathrm{ext}}$$

in the presence of external forces. In the absence of external forces:

$$m\frac{\mathrm{d}v}{\mathrm{d}t} = -\frac{\mathrm{d}m}{\mathrm{d}t}u$$

hence

$$v(t) = v(0) + u \log\left(\frac{m(0)}{m(t)}\right)$$

with m(0), v(0) initial mass and velocity

## 7 Rigid Bodies

**Definition.** A **rigid body** is an extended object that can be considered as a multi-particle system such that the distance between any two particles in the body remains constant, i.e.

$$|\mathbf{r}_i - \mathbf{r}_j| = \text{constant } \forall i, j$$

The possible motion of a rigid body is a superposition of the two basis transformations that are isometries of Euclidean Space, i.e. they preserve distance - i.e. translations and rotations.

#### 7.1 Angular Velocity

**Equation.** Recall from section 5, if a particle is rotating about an axis through O, with angular velocity  $\boldsymbol{\omega}$  then the velocity is  $\dot{\mathbf{r}} = \boldsymbol{\omega} \times \mathbf{r}$ 

with

$$|\dot{\mathbf{r}}| = \omega r_{\perp}$$

where  $r_{\perp}$  is the perpendicular distance to the axis of rotation. If the particle has mass m then the kinetic energy

$$T=\frac{1}{2}m\omega^2 r_{\perp}^2$$



Note that if  $\boldsymbol{\omega} = \boldsymbol{\omega} \mathbf{n}$ , then  $r_{\perp} = |\mathbf{r} \times \mathbf{r}|$ 

$$T = \frac{1}{2}mr_{\perp}^2\omega^2 = \frac{1}{2}I\omega^2$$

 ${\cal I}$  is moment of inertia of particle about axis of rotation.

#### 7.2 Moment of Inertia for a Rigid Body

**Equation.** Consider a rigid body to be mad up of N particles Consider the body to be rotating about an axis through the origin with angular velocity  $\boldsymbol{\omega}$ Then for each particle:

$$\dot{\mathbf{r}}_i = \boldsymbol{\omega} imes \mathbf{r}_i$$

Note that

$$\frac{\mathrm{d}}{\mathrm{d}t} |\mathbf{r}_i - \mathbf{r}_j|^2 = 2(\mathbf{r}_i - \mathbf{r}_j) \cdot (\dot{\mathbf{r}}_i - \dot{\mathbf{r}}_j)$$
$$= 2(\mathbf{r}_i - \mathbf{r}_j) \cdot (\boldsymbol{\omega} \times (\mathbf{r}_i - \mathbf{r}_j))$$
$$= 0$$

Constant with properties of a rigid body, i.e.  $|\mathbf{r}_i - \mathbf{r}_j|$  does not vary in time. Now consider the kinetic energy of body

$$T = \sum_{i=1}^{N} \frac{1}{2} m_i \dot{\mathbf{r}}_i^2$$
$$= \frac{1}{2} I \omega^2$$

with I the moment of inertia of the body for rotation about axis  ${\bf n}$  through origin. Now consider angular momentum:

$$\mathbf{L} = \sum_{i=1}^{N} m_i \mathbf{r}_i \times \underbrace{(\boldsymbol{\omega} \times \mathbf{r}_i)}_{=\dot{\mathbf{r}}_i}$$

Consider  $\boldsymbol{\omega} = \boldsymbol{\omega} \mathbf{n}$ . Then

$$\mathbf{L} = \omega \sum_{i=1}^{N} m_i \mathbf{r}_i \times (\mathbf{n} \times \mathbf{r}_i)$$

**Equation.** Now consider part of  $\mathbf{L}$  which is  $\parallel$  to rotation axis

$$\mathbf{L} \cdot \mathbf{n} = I\omega$$

Component of angular momentum in direction of rotation axis is  $I\omega$ In general **L** is not  $\parallel$  to rotation axis. But we have:

$$\mathbf{L} = \sum_{i=1}^{N} m_i (|\mathbf{r}_i|^2 \boldsymbol{\omega} - (\mathbf{r}_i \cdot \boldsymbol{\omega}) \mathbf{r}_i)$$

(which is a linear function of the vector  $\boldsymbol{\omega}$ ) I.e.

 $\mathbf{L}=I\boldsymbol{\omega}$ 

where I is a matrix like object

**Equation.**  $I_{\alpha\beta}$  is a symmetric tensor:

$$I_{\alpha\beta} = \sum_{i=1}^{N} m_i \{ |\mathbf{r}_i|^2 \delta_{\alpha\beta} - (\mathbf{r}_i)_{\alpha} (\mathbf{r}_i)_{\beta} \}$$

#### 7.3 Calculations of Moments of Inertia

**Equation.** For a solid body we replace mass-weighted sums over particles by mass-weighted integrals. Consider a body occupying a volume V with mass density  $\rho(\mathbf{r})$ Total mass:

$$M = \int_{V} \rho(\mathbf{r}) \,\mathrm{d}V$$

Center of mass position vector

$$\mathbf{R} = \frac{1}{M} \int_{\mathbf{r}} \rho(\mathbf{r}) \, \mathrm{d}V$$

Moment of inertia about axis  ${\bf n}$ 

$$I = \int_{V} \rho(\mathbf{r}) |\mathbf{r}_{\perp}|^{2} \, \mathrm{d}V = \int_{V} \rho(\mathbf{r}) |\mathbf{n} \times \mathbf{r}|^{2} \, \mathrm{d}V$$

**Note.** Use the obvious modifications of these formulae for bodies that correspond to mass distributed over a surface or along a curve, as surface( or area) integrals or as line integrals.

We now calculate I for some very simple examples

**Example.** Uniform thin ring of mass M, radius a, about axis through center of ring and  $\perp$  plane of ring.



For this case we reduce volume integral to a line integral

$$\rho = \text{mass per unit length} = \frac{M}{2\pi a}$$

$$I = Ma^2$$

Every point in the body has  $r_{\perp} = a$ 



**Example.** Uniform this disc of mass M, radius a with axis of rotation through the centre of the disc and  $\perp$  to plane of disc



**Example.** Uniform thin disc, mass M, radius a, axis of rotation is through centre, but in plane of the disc



$$I = \frac{1}{4}Ma^2$$

**Example.** Uniform sphere mass M, radius a, axis of rotation through centre



Use spherical polar co-ordinates  $r, \theta, \phi$  and choose  $\theta = 0$  to correspond to the axis of rotation. Density  $\frac{M}{4\pi a^3/3}$ 

$$I = \frac{2ma^2}{5}$$

**Method.** The following are simple general results that are useful when calculating moments of inertia:

Theorem (Perpendicuar Axes Theorem). For a 2-dimensional body (lamina) then

 $I_z = I_x + I_y$ 

Where  $I_z$  moment of inertia about z axis  $\perp$  lamina  $I_x$ ,  $I_y$  are moments of inertia about respectively the x, y axes which lie in the plane



**Example.** Symmetric case: disc

$$I_z = \frac{1}{2}Ma^2 = 2 \cdot \left(\frac{1}{4}Ma^2\right) = 2I_x = 2I_y$$

 $I_z$  has axis of rotation  $\perp$  plane of disc.  $I_x,\ I_y$  have axis of rotation in plane of disc

**Theorem** (Parallel Axes Theorem). If a rigid body of mass M has moment of inertia  $I_c$  about axis through the centre of mass, then the moment of inertia about a parallel axis a distance d from the axis through the centre of mass is

$$I = I_C + Md^2$$



Choose Cartesian axes with origin O at the centre of mass and z direction  $\parallel$  axus if rotation. Let second axis pas through the point (d, 0, 0). Denote the volume of the body by V. Then

$$I_c = \int_V \rho(x^2 + y^2) \,\mathrm{d}V$$

$$I = \int \rho((x-d)^2) + y^2) \,\mathrm{d}V$$
  
=  $\int \rho(x^2 + y^2) \,\mathrm{d}V - 2 \underbrace{\int \rho x d \,\mathrm{d}V}_{=0} + d^2 \int \rho \,\mathrm{d}V$   
=  $I_c + M d^2$ 

We have second integral 0 as origin id center of mass

**Example.** Uniform thin disc, mass M, radius a. Want to find moment of inertia about axis  $\perp$  plane of disc, through a point on circumference:



#### 7.4 Motion of a Rigid Body

**Remark.** General motion of a rigid body can be described by translation of the centre of mass, following a trajectory  $\mathbf{R}(t)$ , together with a rotation about an axis through the centre of mass. Following section 6.2, we specify the points fixed in the body relative to the mass by writing:

$$\mathbf{r}_i = \mathbf{R} + \mathbf{s}$$

$$\dot{\mathbf{r}}_i = \dot{\mathbf{R}} + \dot{\mathbf{s}}_i$$

and noting:

$$\sum_{i=1}^{N} m_i \mathbf{r}_i = M \mathbf{R} \implies \sum_{i=1}^{N} m_i \mathbf{s}_i = \mathbf{0}$$

**Equation.** If body is rotating about centre of mass with angular velocity  $\omega$ , then

$$\dot{\mathbf{s}}_i = oldsymbol{\omega} imes \mathbf{s}_i$$

and

$$\dot{\mathbf{r}}_i = \mathbf{R} + oldsymbol{\omega} imes \mathbf{s}_i$$

**Equation.** Then kinetic energy (recalling section 6.2) is

$$T = \frac{1}{2}M\dot{\mathbf{R}}^2 + \frac{1}{2}I_C\omega^2$$

where  $I_C$  is the moment about axis  $\parallel \boldsymbol{\omega}$  through the centre of mass

T =translation KE + rotational KE

Shown in section 6.1, for general particle system, we have linear momentum and angular momentum satisfy:

 $\dot{\mathbf{P}} = \mathbf{F}$ 

 $(\mathbf{F} \text{ is total external force})$  and

 $\dot{\mathbf{L}} = \mathbf{G}$ 

(**G** is total external torque)

i.e. for a rigid body there are two equations that determine the translational and rotational motion

**Note.** Can sometimes determine motion by exploiting conservation of energy

**Method.** L and G depend on choice of origin: can take origin to be any point fixed in an inertial frame (considered previously in section 6.1) or we can define L and G with respect to the center of mass – the equation relating them still holds. (can show)

**Example.** Consider motion in a uniform gravitational field, acceleration  $\mathbf{g}$ , which is constant. The total gravitational force and toque acting on a rigid body are the same as those that would act on a particle of mass M located at the centre of mass

(often referred to as the centre of gravity)

To confirm:

$$\mathbf{F} = \sum_{i=1}^{N} \mathbf{F}_{i}^{\text{ext}} = \sum_{i=1}^{N} m_{i} \mathbf{g} = M \mathbf{g}$$

$$\mathbf{G} = M\mathbf{R} \times \mathbf{g}$$

Note. The gravitational toque about the centre of mass is zero

Consider the gravitational potential

$$V^{\text{ext}} = -M\mathbf{R} \cdot \mathbf{g}$$

**Moral.** Consider using gravitational torque = rate of change of angular momentum to work through problems. Also consider energy

## 7.5 Sliding vs Rolling



velocity  $\omega$ 

## 8 Special Relativity

#### 8.1 Postulates of Special Relativity

**Note.** Special Relativity (SR) is based on two postulates. Postulate 1: The laws of physics are the same in all inertial frames. Postulate 2: The speed of light in a vacuum in the same in all inertial frames.

#### 8.2 The Lorentz Transformation



**Equation.** From (1) we have (by subbing in for vt'):

$$' = \gamma \left( t - \frac{vx}{c^2} \right) \tag{3}$$

(2) and (3) define the Lorentz transformation (or 'Lorentz Boost') Straightforward to show from (2) and (3) that:

$$x = \gamma(x' + vt')$$
$$t = \gamma\left(t' + \frac{vx}{c^2}\right)$$

(note that the sign of v has simply been swapped)

Note. a general property of Lorentz transformations is:

$$c^{2}t'^{2} - r'^{2} = c^{2}t^{2} - x^{2} - y^{2} - z^{2} = c^{2}t^{2} - r^{2}$$

So

 $r' = ct \iff r = ct$ 

Note quantity is invariant under Lorentz transformations

#### 8.3 Space-time Diagrams

**Method.** We can also draw axes of frame S' which is moving at speed v relative to S. t' axis: corresponds to x' = 0 hence to x = vt or

$$x = \frac{v}{c} \cdot ct$$

x'axis: corresponds to t' = 0 hence



Equation. Comparison of velocities.

Consider a particle moving with constant velocity u' in S', with S' is travelling at velocity v, relative to S.

$$u = \frac{u' + v}{1 + \frac{u'v}{r^2}}$$

Formula for composition of velocities (inverse by swapping sign of v)

(inverse by swapping sign of v)

In the limit  $u',v\ll c,$  we recover Galilean transformation result

#### 8.4 Simultaneity, Causality and Consequences

#### 8.4.1 Simultaneity



**Remark.** Events that are simultaneous in S' have the same value of t' i.e. they lie on lines t - vx/c constant.

event  $P_2$  occurs before  $P_1$  in 'S'.

Note.

$$P_1(x_1, ct), P_2(x_2, ct) \implies t_1' = \gamma \left(t - \frac{vx_1}{c^2}\right), t_2' = \gamma \left(t - \frac{vx_2}{c^2}\right)$$

Hence  $t'_{2} < t'_{1}$  if  $x_{2} > x_{1}$ 

Warning. Simultaneity IS frame dependent

#### 8.4.2 Causality

Different observers may disagree on the time ordering of events, but we can construct a viewpoint where cause and effect are consistent.

Lines of simultaneity cannot be inclined at more than 45° to x axis (because |v| < c)



#### 8.4.3 Time Dilation

**Method.** A clock that is stationary in S' ticks at time intervals  $\delta t'$ . What is the time interval between ticks in S? Lorentz transformation:

$$t = \gamma(t' + \frac{vx'}{c^2})$$

hence

 $\delta t = \gamma \delta t'$ 

moving clocks run slowly.

**Definition. Proper time** is time measured in the rest frame of a moving object

#### 8.4.4 Length Contraction



Definition. Proper length is the length measured in the rest frame of the object

#### 8.5 Geometry of Space-time

#### 8.5.1 Invarianvce Interval

**Equation.** Consider two events P, Q with space time co-ordinates  $(ct_1, x_1)$ ,  $(ct_2, x_2)$ Time separation

$$\Delta t = t_1 - t_2$$

Space separation

$$\Delta x = x_t - x_2$$

Invariance interval between P and Q is defined as

$$\left(\Delta s\right)^2 = c^2 \Delta t^2 = \Delta x^2$$

All observers in inertial frames agree on the value of  $(\Delta s)^2$ . (Exercise: show above from Lorentz transformation) In 3 space dimensions, define the invariant interval as

$$\Delta s^2 = c^2 \Delta t^2 - \Delta x^2 - \Delta y^2 - \Delta z^2$$

If the separation between P and Q is very small, then we have an expression for the infinitesmval invariant interval.

$$ds^{2} = c^{2} dt^{2} - (dx^{2} + dy^{2} + dz^{2})$$

Space-time is topologically equivalent to  $\mathbb{R}^4$ , when endowed with a distance measure  $ds^2$ . (though  $ds^2$  is not positive definite.)

This is Minkowski space time - sometimes described as a space of 1 + 3 dimensions.

## **Definition.** Events with $ds^2 > 0$ are **time-like separated**

There is a frame of reference such that the two events occur at the same space position (but at different times).



The event Q lies in either the forward or the backward light cone of P. (The time-ordering of P and Q is agreed by all observers)





Q lies outside the forward and backward light cone of P. (Which of P or Q occurs at earlier time is observer dependent.)

**Definition.** If  $ds^2 = 0$  then the events are **light-like separated** and one lies on the light cone of the other.



**Warning.**  $ds^2 = 0$  does NOT imply that P and Q are the same event. (Our metric does not have the 'usual' metric properties.)

#### 8.5.2 The Lorentz Group

**Method.** The co-ordinates of an event P in frame S can be written as a 4-vector (i.e. a 1 component vector) - X.

$$X = \begin{bmatrix} ct \\ x \\ y \\ z \end{bmatrix}$$

we use superscripts to denote index (write  $X^{\mu}$ )

$$X^1 = ct, \ X^2 = x, \ X^3 = y, \ X^3 = z$$

Equation. Invariance interval between P and origin O can be written as an inner product

$$X \cdot X = X^T \eta X = X^\mu \eta_{\mu\nu} X^\mu$$

where

$$\eta = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & -1 & 0 & 0 \\ 0 & 0 & -1 & 0 \\ 0 & 0 & 0 & -1 \end{bmatrix}$$

the Minkowski metric. We have

$$X \cdot X = c^2 t^2 - x^2 - y^2 - z^2$$

**Note.** 4 vectors with  $X \cdot X > 0$  are time like 4 vectors with  $X \cdot X < 0$  are space like 4 vectors with  $X \cdot X = 0$  are light like

**Remark.** The Lorentz transformation is a linear transformation that takes the components X in S to be components X' in S'

**Method.** The Lorentz transformation can be represented by a  $4 \times 4$  matrix  $\Lambda$ .

$$X' = \Lambda X$$

X are the components in SX' are the components in S'Lorentz transformations can be defined as the set of  $\Lambda$  that leave the inner product  $X \cdot X$  unchanged, i.e.

$$X \cdot X = X' \cdot X'$$
 for all  $\Lambda$ 

implying

$$\Lambda^T \eta \Lambda = \eta$$

(\*)

**Example.** Two classes of possible  $\Lambda$  are

$$\Lambda = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & & \\ 0 & & R \\ 0 & & & \end{bmatrix}$$

with

with

$$R^T R = I$$

i.e.  ${\cal R}$  is spacial reflection/ rotation. And:

$$\Lambda = \begin{bmatrix} \gamma & -\gamma\beta & 0 & 0\\ -\gamma\beta & \gamma & 0 & 0\\ 0 & 0 & 1 & 0\\ 0 & 0 & 0 & 1 \end{bmatrix}$$
$$\beta = \frac{v}{c}$$
$$\gamma = \left(1 - \frac{v^2}{c^2}\right)^{-1/2}$$

Lorentz boost, with velocity in x direction (in this example).

**Definition.** The set of all  $\Lambda$  satisfying (\*) is called the **Lorentz group** O(1,3). It includes time reversals and spatial reflections.

The subgroup with det  $\Lambda = 1$  is the **proper Lorentz group** SO(1,3), which includes composition of time reversal and spacial reflection.

The subgroup that preserves the direction of time and spatial circulation is the **restricted Lorentz** group  $SO^+(1,3)$  - generated by rotations and boosts (in arbitrary directions).

#### 8.5.3 Rapidity

This is a way of labelling Lorentz transformations

**Method.** Focus on  $2 \times 2$  submatrix operations on (ct, x). Write:

$$\Lambda[\beta] = \begin{bmatrix} \gamma & -\gamma\beta \\ -\gamma\beta & \gamma \end{bmatrix} \ \gamma = (1-\beta^2)^{-1/2}$$

(boost in *x*-direction). Consider combining two boosts:

$$\Lambda[\beta_1]\Lambda[\beta_2] = \Lambda\left[\frac{\beta_1 + \beta_2}{1 + \beta_1\beta_2}\right]$$

Check - note relation to velocity transformation law. Recall that for spacial rotations

$$R(\theta) = \begin{bmatrix} \cos\theta & \sin\theta \\ -\sin\theta & \cos\theta \end{bmatrix}$$

with

$$R(\theta_1)R(\theta_2) = R(\theta_1 + \theta_2)$$
 – composition law

For Lorentz boosts, define  $\varphi$  such that  $\beta = \tanh \varphi$ 

$$(\implies \gamma = \cosh \varphi, \ \gamma \beta = \sinh \varphi)$$

Then the composition law can be

$$\Lambda(\varphi_1)\Lambda(\varphi_2) = \Lambda(\varphi_1 + \varphi_2)$$

Suggests that Lorentz boosts can be regarded as 'hyperbolic rotations' in space-time

## 8.6 Relativistic Kinematics

Method. A particle moves along a trajectory  $\mathbf{x}(t)$ . Its velocity is

$$\frac{\mathrm{d}}{\mathrm{d}\mathbf{x}} = \mathbf{u}(t)$$

Path in space-time is parametrised by t. But both  $\mathbf{x}$  and t vary under Lorentz transformation – describing the path in a different frame will be difficult. Proper time: First consider a particle at rest in S' with

 $\mathbf{x}' = \mathbf{0}$ 

The invariant interval between points on its world line is

$$\Delta s^2 = c^2 \Delta t'^2$$

Define proper time  $\tau$  such that

$$\Delta \tau = \frac{1}{c} \Delta s$$

This is the change in time experienced in the rest frame of the particle. But the equation

$$\Delta \tau = \frac{1}{c} \Delta s$$

holds in all frames because  $\Delta s$  is Lorentz invariant. Note that  $\Delta s$  is real in all frames - the interval is timelike.

The world line of the particle can now be parametrised in terms of  $\tau$ 



Method (continued). Infinitesimal changes are related by:

$$\mathrm{d}\tau = \left(1 - \frac{u^2}{c^2}\right)^{1/2} \,\mathrm{d}t$$

Hence

$$\frac{\mathrm{d}t}{\mathrm{d}\tau} = \gamma_{\mathbf{u}} \text{ with } \gamma_{\mathbf{u}} = \frac{1}{\sqrt{1 - \frac{u^2}{c^2}}}$$

Total time observed by particle moving along its world line

$$T = \int \mathrm{d}\tau = \int \frac{\mathrm{d}t}{\gamma_{\mathbf{u}}}$$

**Definition** (4-velocity). **Position 4-vector** of a particle is

$$X(\tau) = \begin{bmatrix} ct(\tau) \\ \mathbf{x}(\tau) \end{bmatrix}$$

The **4-velocity** is defined by

$$U = \gamma_{\mathbf{u}} \begin{bmatrix} c \\ \mathbf{u} \end{bmatrix}$$

where

$$\mathbf{u} = \frac{\mathrm{d}\mathbf{x}}{\mathrm{d}t}$$

Notation. Another possible notation:

$$(ct, \mathbf{x}), (\gamma_{\mathbf{u}} c, \gamma_{\mathbf{u}} \mathbf{u})$$
  
position 4-vector 4-velocity

**Equation.** If frames S and S' are such that X and X' are corresponding components of position vector, then

 $X' = \Lambda X$ 

Correspondingly

 $U' = \Lambda U$ 

relates the components of the 4-velocity

(Any quantity whose components transform according to this rule is a 4-vector.) U is a 4-vector because X is a 4-vector and  $\tau$  is an invariant (i.e.  $\frac{dX}{d\tau}$  is a 4-vector). But e.g.  $\frac{dX}{dt}$  is not a 4-vector because t is not an invariant. The scalar product  $U \cdot U$  is invariant under Lorentz transformations, i.e.

$$U \cdot U = U' \cdot U'$$

In the rest frame of a particle moving with 4-velocity U,

$$U = \begin{bmatrix} c \\ \mathbf{0} \end{bmatrix}$$

hence

$$U \cdot U = c^2$$

In any other frame

$$U \cdot U = c^2 = \gamma^2 (c^2 - \mathbf{u}^2)$$

as expected

#### 8.6.1 Transformations of Velocities as 4-Vectors

**Note.** We have previously seen that the transformation law for velocities in SR is not simply the addition rule implied by Galilean transformations.



**Definition.** There is change in perceived angle of velocity, – '**aberration**' (due to the motion of the observer).

applies when u = c, i.e. aberration applies to light rays. (Apparent direction of stars changes as a result of the Earth's orbital motion.)

#### 8.7 Relativistic Physics

**Definition.** 4-momentum of a particle with mass m and with 4-velocity U is

$$P = mU = m\gamma_{\mathbf{u}} \begin{bmatrix} c \\ \mathbf{u} \end{bmatrix}$$

(has 4 components  $P^0$ ,  $P^1$ ,  $P^2$ ,  $P^3$ )

For P to be a 4-vector, then m must be an invariant.

m is the rest mass, i.e. mass measured in the rest frame of the particle.

4 momentum of a system of particles is the sum of the 4-momenta of the individual particles – conserved in absence of external forces.

**Equation.** Spatial components of  $P(\mu = 1, 2, 3)$  are relativistic 3 momentum

 $\mathbf{p} = \gamma_{\mathbf{u}} m \mathbf{u}$ 

Same as for Newtonian physics except that mass m is modified to  $\gamma_{\mathbf{u}}m$ .

**Definition.** We interpret  $\gamma_{\mathbf{u}}m$  as the '**apparent mass**'. Note that  $|\mathbf{p}|$  and the apparent mass  $\rightarrow \infty$  as  $|\mathbf{u}| \rightarrow c$ .

Moral. What does  $P^0$  represent?

$$P^{0} = \gamma_{\mathbf{u}}mc = \frac{mc}{\sqrt{1 - \mathbf{u}^{2}/c^{2}}} = \frac{1}{c} \left(mc^{2} + \frac{1}{2}m\mathbf{u}^{2} + \dots\right)$$

Natural interpretation of  $P^0$  is an energy.

$$P = \begin{bmatrix} E/c \\ \mathbf{p} \end{bmatrix} \text{ with } E = \gamma_{\mathbf{u}} mc^2 = mc^2 + \frac{1}{2}m\mathbf{u}^2 + \dots$$

Note that  $E \to \infty$  as  $|\mathbf{u}| \to c$  (if  $m \neq 0$ ). (*P* is sometimes called the energy-momentum 4-vector)

**Equation.** For a stationary particle with rest mass m, we have that

$$E = mc^2$$

Implication is that mass if a form of energy. Energy of a moving particle

$$E = mc^{2} + \underbrace{(\gamma_{\mathbf{u}} - 1)mc^{2}}_{\text{Relativistic K.E.}}$$

Relativistic K.E. is a generalisation of the Newtonian formula. Since

$$P \cdot P = \frac{E^2}{c^2} - |\mathbf{p}|^2$$

is Lorentz invariant, we have

$$P \cdot P = m^2 c^2$$

hence

$$\frac{E^2}{c^2} = p^2 + m^2 c^2$$

(important relation between momentum and energy)

**Note.** SUMMARY:

• 4-momentum

$$P = \begin{bmatrix} E/c \\ \mathbf{p} \end{bmatrix}$$

- E is relativistic energy
- $P \cdot P$  is invariant

$$P \cdot P = E^2/c^2 - |\mathbf{p}|^2 = m^2 c^2$$

(m is rest mass)

• Mass is a form of energy – can convert into KE and vice versa.

#### 8.7.1 Massless Particles

**Definition.** Massless particles are particles with zero rest mass (m = 0), e.g. photons.

**Equation.** These particles can have non-zero momentum and non-zero energy, because they travel at the speed of light.

 $(\gamma_{\mathbf{u}} = \infty).$ 

In this case  $P \cdot P = 0$ .

These particles are travelling along light-like trajectories and therefore they have no proper time (can't transform to rest frame).

Energy and (3-)momentum for such particles satisfies:

$$\frac{E^2}{c^2} = |\mathbf{p}|^2$$

i.e.

$$E = |\mathbf{p}|c$$

Then

$$P = \frac{E}{c} \begin{bmatrix} 1\\ \mathbf{n} \end{bmatrix}$$

with  $\mathbf{n}$  unit vector in direction of travel.

8.7.2 Newton's 2<sup>nd</sup> Law in Special Relativity

Equation. Law has the form

$$\frac{\mathrm{d}P}{\mathrm{d}\tau} = F$$

where F is a 4-vector – the 4-force

**Equation.** The relation between 4-force F and 3-force  $\mathbf{F}$  is

$$F = \gamma_{\mathbf{u}} \begin{bmatrix} \mathbf{F} \cdot \mathbf{u}/c \\ \mathbf{F} \end{bmatrix}$$

Thus

$$\frac{\mathrm{d}E}{\mathrm{d}\tau} = \gamma_{\mathbf{u}} \frac{\mathbf{F} \cdot \mathbf{u}}{c}, \ \frac{\mathrm{d}\mathbf{p}}{\mathrm{d}\tau} = \gamma_{\mathbf{u}} \mathbf{F}$$

hence

$$\frac{\mathrm{d}E}{\mathrm{d}t} = \frac{\mathbf{F} \cdot \mathbf{u}}{c}, \ \frac{\mathrm{d}\mathbf{p}}{\mathrm{d}t} = \mathbf{F}$$

Equivalently, for particle of rest mass m

$$F = mA$$
 with  $A = \frac{\mathrm{d}U}{\mathrm{d}\tau}$ , the 4-acceleration

Note. Equate 4-momentum to work through problems. Can work in centre of momentum frame.